

Improved Performance of DFIG-generators for Wind Turbines Variable-speed

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ABSTRACT

The following article presents the control of the power generated by the Doubly Fed Induction Generator, integrated into the wind system, whose rotor is linked to the power converters (Rotor Side Converter (RSC) and Grid Side Converter (GSC)) interfaced by the DC-BUS and connected to the grid via a filter (Rf, Lf) in order to obtain an optimal power to the grid and to ensure system stability. The objective of this study is to understand and to make the comparison between Sliding mode Control technique and the Flux Oriented Control in order to control the Doubly Fed Induction Generator powers exchanged with the grid, it also aims at maintaining the DC-BUS voltage constant and a unit power factor at the grid connection point. The results of simulation show the performance of the Sliding mode Control in terms of monitoring, and robustness with regard to the parametric variations, compared to the Flux Oriented Control. The performance of the systems was tested and compared with the use of MATLAB/Simulink software

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1. INTRODUCTION

Recently, energy demands have grown strongly, while sources of sickle fuels have decreased enormously. For this reason, the development and exploitation of renewable energies have grown extremely. Among these sources, we find the wind energy which has been the subject of several researches in electrical and electromechanical engineering to improve the quality of the produced energy. Currently, wind turbine systems with variable speed based on the Doubly Fed Induction Generator as shown in Figure 1 are the most commonly used in wind farms because they allow the operation on a speed range of $\pm 30\%$ around the synchronous speed, thanks to the dimensioning of the three-phase static converters for a part of the nominal power $\pm 30\%$, which makes it possible to reduce losses in electronic power components and the overall system increase. Therefore, not only does energy production depend on the way in which the converters are controlled but also on the DFIG generator. The rotor side converter "RSC" controls the reactive and active power of the DFIG whereas the grid side converter "GSC" controls the DC-LINK voltage and power factor. The major problem of the Doubly Fed Induction Generator is that it is characterized by a non-linear, multi-variable mathematical model with a heavy coupling between input variables (i.e.) it is not possible to independently control the voltage or the current [1]. In order to control this machine, many works of implementation which use approaches based on linear models have been applied, but the approach by linear controller has rapidly showed its limits. That is why this researches are turned towards the nonlinear techniques to increase the robustness and the precision of the systems to be controlled. The most known one is the Sliding Mode Control identified by its insensitivity to variations in internal and external parameters, stability, simplicity and very low response times [2] [3]. The principle of this technique consists in bringing

the system state trajectory towards the sliding surface and to switch it by means of appropriate switching logic around it to the equilibrium point[4] [5].

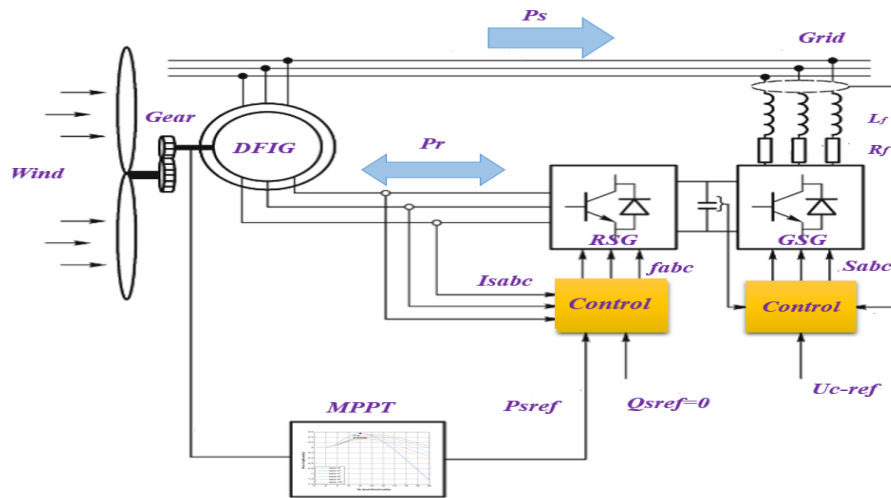


Figure 1. Global wind energy conversion system

In this paper, the modeling of the wind system as well as the MPPT strategy will be presented, in addition to a study of the DFIG model in the Park referential. After that, we will discuss the theoretical part of sliding mode control, followed by its application to the RSC to independently control the active and reactive powers generated by the DFIG, and to the GSC to ensure a DC-BUS voltage and ensure sinusoidal currents in the side of the grid. Finally, we will present and examine the simulation results using MATLAB / SIMULINK so as to make a comparison with the linear control by Flux Oriented Control with the objective of improving the wind system performance.

2. THE MECHANICAL PART OF THE SYSTEM

2.1. Mathematical Model Of The Turbine

The kinetic power of the wind according to Bernoulli's theorem is given as a function of the area swept by the turbine blades(S), the air density (ρ) and the wind speed (V) by the following Equation [6-9]:

$$P_v = \frac{\rho S V^3}{2} \quad (1)$$

The wind turbine can convert a percentage of the kinetic wind power into a mechanical power. The aerodynamic power appearing at the rotor of the turbine is written as follows [9-11]:

$$P_{aero} = C_p(\lambda, \beta) \cdot \frac{\rho \pi R^2 V^3}{2} \quad (2)$$

C_p presents the power coefficient as shown in Figure 2, which depends on the characteristic of the wind turbine (λ and β). This later is approximated by the following Equation [10], [8]:

$$C_p(\lambda, \beta) = C_1 \left(\frac{C_2}{\lambda} - C_3 \beta - C_4 \right) \exp \left(-C_5 \left(\frac{1}{(\lambda + 0.08\beta)} - \frac{0.035}{\beta^3 + 1} \right) \right) + C_6 \lambda \quad (3)$$

From Figure 2, we can note that the power coefficient C_p reaches its maximum 0.5506 for a speed ratio $\lambda_{opt}=8$ and $\beta=0^\circ$. The speed ratio λ presents the relationship between the wind speed and that of the turbine Ω_t . Its expression is given as follows [8-9]:

$$\lambda = R \cdot \frac{\Omega_t}{V} \quad (4)$$

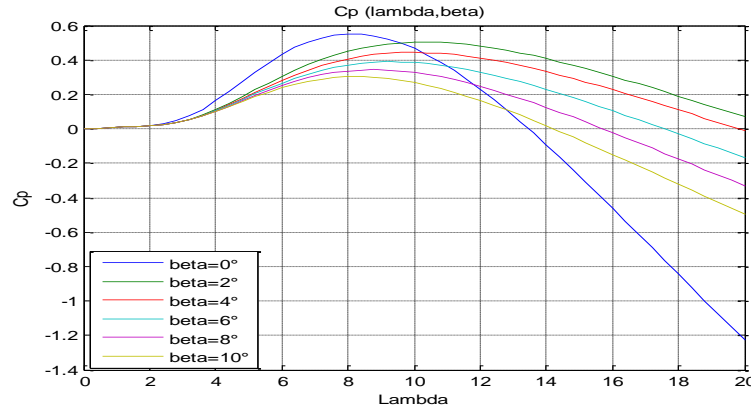


Figure 2. Power coefficient in function of λ for diverse values of β

2.2. Gerbox

The gearbox allows to adapt the slow turbine speed to that required by the generator. It is modeled by the following equation system [5-6]:

$$\begin{cases} \Omega_t = \frac{\Omega_{mec}}{G} \\ C_g = \frac{C_{aero}}{G} \end{cases} \quad (5)$$

2.3. Dynamic Equation Of The Shaft

The fundamental equation of the dynamics makes it possible to define the progress of the mechanical speed Ω_{mec} [8] [10]:

$$C_{mec} = J \frac{d\Omega_{mec}}{dt} = C_g - C_{em} - f \cdot \Omega_{mec} \quad (6)$$

With: C_{mec} : Mechanical torque exerted on the rotor shaft of the wind turbine. C_{em} : Electromagnetic torque, $f \cdot \Omega_{mec}$: The torque of viscous friction. J : The total inertia consists of the turbine inertia J_t and the generator inertia J_g given by:

$$J = \frac{J_t}{G} + J_g \quad (7)$$

2.4. Maximum Power Point Tracking Strategy

In order to extract the maximum power from the wind, we need an algorithm acting on the set point variables, to have a good efficiency of the device. For this reason, we applied a technique of Maximum Power Point Tracking Strategy. This technique consists in imposing a torque of reference so as to permit the DFIG to turn at a regulating speed to ensure an optimal operating point of power extraction [3]. That is why the speed ratio λ must be kept its optimum value ($\lambda = \lambda_{opt}$) over a certain range of wind speed, furthermore, the power coefficient would be maintained at its maximum value ($C_{pmax} = C_p$) [3,12]. In this case, the aerodynamic couple will have as an expression:

$$C_{aero} = C_{p_max}(\lambda, \beta) \cdot \frac{\rho \cdot \pi R^2 \cdot V^3}{2 \cdot \Omega_t} \quad (8)$$

The reference torque at the output of the multiplier becomes:

$$C_{g_{ref}} = \frac{1}{G} \cdot C_{aero} \quad (9)$$

We know that the fundamental equation of dynamics is given by:

$$C_{mec} = J \frac{d\Omega_{mec}}{dt} = C_g - C_{em} - C_f \quad (10)$$

We assumed that the generator rotational speed is fixed during the study period, and we neglected the effect of the viscous torque. The reference electromagnetic torque can be expressed by :

$$C_{emref} = C_g \quad (11)$$

The estimated wind speed can be written as suggested:

$$V_{est} = R \cdot \frac{\Omega_{t-est}}{\lambda_{est}} \quad (12)$$

The expression of the reference electromagnetic torque can be written as indicated:

$$C_{emref} = \frac{C_{p_max}}{\lambda_{opt}^3} \cdot \frac{\rho \cdot \pi \cdot R^5}{2} \cdot \frac{\Omega_{mec}^2}{G^3} \quad (13)$$

Based on these equations, we can create the following diagram of the mechanical part of the wind system and the MPPT strategy without speed measurement shown in Figure 3 [5]:

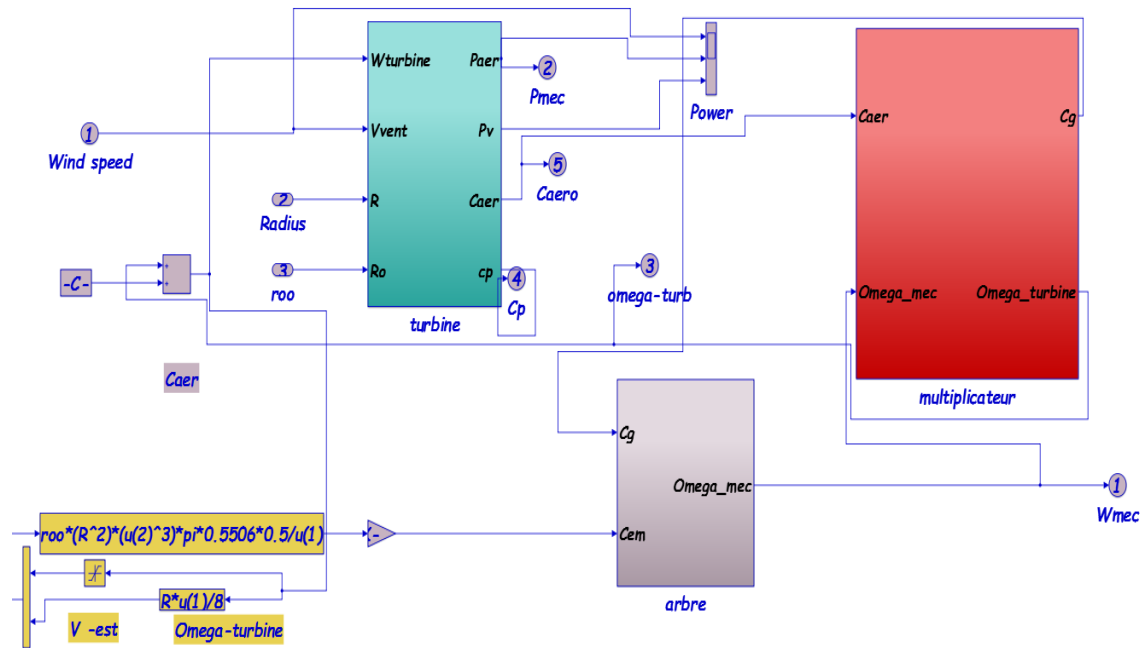


Figure 3. The MPPT strategy without speed measurement

3. THE ELECTRICAL PART OF THE SYSTEM

3.1. The DFIG Model

The Doubly Fed Induction Generator DFIG model using Park transformation is presented by the following Equations [6-13][23-24]. Stator and rotor voltages in the reference of Park:

$$\begin{cases} V_{sd} = R_s \cdot I_{sd} + \frac{d\phi_{sd}}{dt} - \omega_s \cdot \phi_{sq} \\ V_{sq} = R_s \cdot I_{sq} + \frac{d\phi_{sq}}{dt} - \omega_s \cdot \phi_{sd} \\ V_{rd} = R_r \cdot I_{rd} + \frac{d\phi_{rd}}{dt} - \omega_r \cdot \phi_{rq} \\ V_{rq} = R_r \cdot I_{rq} + \frac{d\phi_{rq}}{dt} + \omega_r \cdot \phi_{rd} \end{cases} \quad (14)$$

Stator and rotor Flux in the reference of Park:

$$\begin{cases} \Phi_{sd} = L_s \cdot I_{sd} + M \cdot I_{rd} \\ \Phi_{sq} = L_s \cdot I_{sq} + M \cdot I_{rq} \\ \Phi_{rd} = L_r \cdot I_{rd} + M \cdot I_{sd} \\ \Phi_{rq} = L_r \cdot I_{rq} + M \cdot I_{sq} \end{cases} \quad (15)$$

Electromagnetic Torque:

$$C_{em} = p(\Phi_{sd} \cdot I_{sq} - \Phi_{sq} \cdot I_{sd}) \quad (16)$$

Active and reactive DFIG powers:

$$\begin{cases} P_s = V_{sd} \cdot I_{sd} + V_{sq} \cdot I_{sq} \\ Q_s = V_{sq} \cdot I_{sd} - V_{sd} \cdot I_{sq} \end{cases} \quad (17)$$

3.2. Filter (R,L) Model

The GSC Converter is connected to the DC-BUS and the Grid via a filter (Rf, Lf). It has two roles: sustaining the DC- BUS voltage constance regarding the amplitude and the rotor power direction and maintaining a unit power factor at the link point to the grid [14].The filter model in the referential (d, q) is given by Equation 18:

$$\begin{cases} v_{df} = -R_f I_{df} - L_f \frac{dI_{df}}{dt} + \omega_s L_f I_{qf} + v_{sd} \\ v_{qf} = -R_f I_{qf} - L_f \frac{dI_{qf}}{dt} - \omega_s L_f I_{df} + v_{sq} \\ P_f = v_{df} I_{df} + v_{qf} I_{qf} \\ Q_f = v_{qf} I_{df} - v_{df} I_{qf} \end{cases} \quad (18)$$

4. FLUX ORIENTED CONTROL TECHNIQUE

4.1. Flux Oriented Control Principle

The principle of Flux Oriented Control FOC consists in orienting the flux along one of the axes in order to make the functioning of the induction machine identical to that of the DC machine separately excited[5] [6] .In this case we have chosen to place the stator flux on the axis”d” as shown in Figure 4.

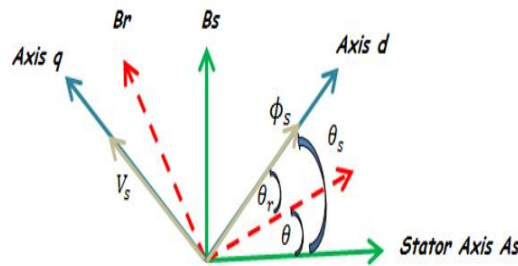


Figure 4. Stator flux orientation

As a result of this orientation, the quadrature part of the stator flux is null and the direct component is equal to the total stator flux. So, we get the following equation:

$$\begin{cases} v_{sq} = v_s \\ v_{sd} = 0 \end{cases} \quad (19)$$

4.2. Application of the Flux Oriented Control FOC to the RSC

According to the Flux Oriented Control principle, the previous equations become as follows:

$$\begin{cases} V_{sd} = 0, V_{sq} = V_s = \omega_s \cdot \phi_s \\ P_s = -V_s \cdot \frac{M}{L_s} \cdot I_{rq} \\ Q_s = \frac{V_s^2}{\omega_s \cdot L_s} - V_s \cdot \frac{M}{L_s} \cdot I_{rd} \\ V_{rq} = (R_r + S \cdot \left(L_r - \frac{M^2}{L_s} \right)) \cdot I_{rq} + \omega_s \cdot g \cdot \left(L_r - \frac{M^2}{L_s} \right) \cdot I_{rd} + \frac{g \cdot M \cdot V_s}{L_s} \\ V_{rd} = (R_r + S \cdot \left(L_r - \frac{M^2}{L_s} \right)) \cdot I_{rd} - \omega_s \cdot g \cdot \left(L_r - \frac{M^2}{L_s} \right) \cdot I_{rq} \end{cases} \quad (20)$$

4.3. Application of the Flux Oriented Control FOC to the GSC

In order to obtain a robust wind turbine system with variable speed, we need to control the GSC, so that the DC-BUS voltage will kept a constant value to ensure a good DC-BUS voltage regulation and a unit power side of the grid. By using Flux Oriented technique FOC, the previous equations are as follows:

$$\begin{cases} v_{df} = -R_f I_{df} - L_f \frac{dI_{df}}{dt} + \omega_s L_f I_{qf} \\ v_{qf} = -R_f I_{qf} - L_f \frac{dI_{qf}}{dt} - \omega_s L_f I_{df} + v_s \\ P_f = v_s I_{qf} \\ Q_f = -v_s I_{df} \end{cases} \quad (21)$$

Then, we can set up a filter current controller (R_f, L_f) with a decoupling between the magnitudes which allows their independent control. The relation between the Filter Currents and the Filter Voltages is given by the following equation:

$$\begin{cases} v_{df} = -(R_f + S \cdot L_f) \cdot I_{df} + \omega_s L_f I_{qf} \\ v_{qf} = -(R_f + S \cdot L_f) \cdot I_{qf} - \omega_s L_f I_{df} + v_s \end{cases} \quad (22)$$

5. SLIDING MODE CONTROL SMC

5.1. Sliding Mode Control Principle

The variable structure system (VSS), is a system whose structure changes during its operation. It is characterized by the choice of a structure and a switching logic. This latter allows the system to switch from one structure to another at any time by the choice of a function, which separates the state space into two parts as shown in Figure 5, and an appropriate switching logic. The Sliding mode technique is a special case of (VSS). It consists of forcing the system state trajectory to attain a hyper surface in finite time and then stay there. This latter presents a relation between the system state variables which defines a differential equation, and consequently, determines the system dynamics if it remains on the hyper surface. The evolution of a system subject to a control law no longer depends on the system or the external disturbances, but on the properties of the hyper surface. The system will therefore be robust not only to uncertainties (specific to the system) and disturbances (external to the system) but will be totally insensitive because they are completely rejected by the control [15].

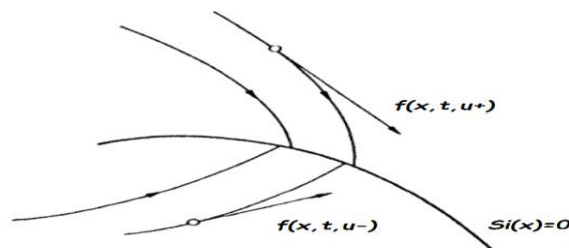


Figure 5. Sliding mode technique principle

The plan of the sliding mode control algorithm is mainly carried out in three steps defined by [5]:

Sliding surface

The sliding surface proposed by SLOTIME is given by [16,17] :

$$s(x,t) = \left(\frac{\partial}{\partial t} + \lambda \right)^{(n-1)} e(x) \quad (23)$$

With: $e(x)$: deviation of the variable to be adjusted $e(x) = x_{ref} - x$, λ : positive constant. n : presents the number of times it derives the output to make the command appear.

Convergence condition

To ensure the sliding mode control stability, LYAPUNOV proposes a positive function $V(x)$ which guarantees the nonlinear system stability and the attraction of the state variable to its reference value. This latter is defined as follows [3,5]:

$$V(x) = \frac{1}{2} s(x)^2 \quad (24)$$

The function $V(x)$ derivation must be negative, so that the variable $S(x, t)$ can tend towards zero, as shown in the following equation :

Control Law

$$\dot{V}(x) = s(x)\dot{s}(x) < 0 \quad (25)$$

The command (u) is a variable structure command given by [19]:

$$u = \begin{cases} u^+(x) & \text{if } s(x,t) > 0 \\ u^-(x) & \text{if } s(x,t) < 0 \end{cases} \quad (26)$$

The control (u) of discontinuous nature, will oblige the system trajectories to reach the sliding surface and to remain on that surface in vicinity of this latter, regardless of the disturbances. According to the equation system (26), we can note that the command (u) is not defined for $S=0$. For this reason, FILLIPOV [19] and UTKIN [20] propose a method called the equivalent command [5,18], shown in the following Figure 6.

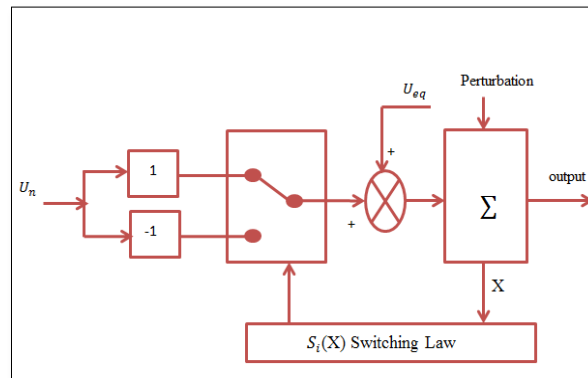


Figure 6. the equivalent command

From Figure 6, we can see that the equivalent command is expressed by;

$$u = u_{eq} + u_n \quad (27)$$

U_{eq} : presents the equivalent part of the control used to keep on the sliding surface, the variable which must be controlled:

$$\dot{S}(x) = S(x) = 0 \quad (28)$$

u_n : The stabilizing control allows us to ensure a convergence as well as the robustness regarding the disturbances.

$$U_n = K \cdot \text{sign}(S(x)) \quad (29)$$

The main disadvantage of this control type is the phenomenon known by « CHATTERING » [3,5] which appears in steady state as a high frequency oscillation around the equilibrium point, because of the very discontinuous nature of the sign function. To solve this problem, several techniques have been applied, among them we find the one proposed by J.SLOTINE which consists of replacing the “SIGN” function by the “SAT” function given by :

$$\text{Sat}(S) = \begin{cases} \text{Sat}(S) = 1 & \text{if } S > \varepsilon \\ \text{Sat}(S) = -1 & \text{if } S < -\varepsilon \\ \text{Sat}(S) = \frac{S}{\varepsilon} & \text{if } |S| < \varepsilon \end{cases} \quad (30)$$

5.2. RSC Control Using Sliding Mode Control

Considering the sliding surface proposed by SLOITINE and taking $n=1$, we obtain:

$$\begin{cases} S(P_s) = e_1 = P_{sref} - P_s \\ S(Q_s) = e_2 = Q_{sref} - Q_s \end{cases} \quad (31)$$

With: Q_{sref} and P_{sref} are the reactive and active powers references. The derivative of sliding surface is given by :

$$\begin{cases} \dot{e}_1 = \dot{P}_{sref} + \frac{V_s M}{L_s} \left(\frac{V_{rqeq} + V_{rqn}}{L_r \sigma} - \frac{R_r}{L_r \sigma} I_{rq} - \omega_r I_{rd} - \omega_r V_s \frac{M}{L_r L_s \omega_s \sigma} \right) \\ \dot{e}_2 = \dot{Q}_{sref} + V_s \frac{M}{L_s} \left(\frac{V_{rdq} + V_{rdn}}{L_r \sigma} - \frac{R_r}{L_r \sigma} I_{rd} + \omega_r I_{rq} \right) \end{cases} \quad (32)$$

During the sliding and permanent mode, we have:

$$e_{(1,2)} = 0, \dot{e}_{(1,2)} = 0, V_{rdn} = V_{rqn} = 0 \quad (33)$$

The expression of the equivalent command becomes:

$$\begin{cases} V_{rdq} = -\frac{L_r L_s \sigma}{M V_s} \dot{e}_2 + R_r I_{rd} - \omega_r L_r \sigma I_{rq} \\ V_{rqeq} = -L_r L_s \frac{\sigma}{V_s M} \dot{e}_1 + R_r I_{rq} + L_r \omega_r \sigma I_{rd} + \omega_r M \frac{V_s}{L_s \omega_s} \end{cases} \quad (34)$$

Therefore, the stabilizing control is given by the following equation:

$$\begin{cases} V_{rdn} = K_d \text{sat}(e_2) \\ V_{rqn} = K_q \text{sat}(e_1) \end{cases} \quad (35)$$

5.3. GSC Control Using Sliding Mode Control

We consider the SLOITINE's sliding mode surface. For $n=1$, we obtain :

$$\begin{cases} S(P_f) = e_3 = P_{fref} - P_f \\ S(Q_f) = e_4 = Q_{fref} - Q_f \end{cases} \quad (36)$$

Where: Q_{fref} and P_{fref} are the reactive and active powers references. The derivative of the sliding surface is given by:

$$\begin{cases} \dot{\mathcal{S}}_1 = \dot{I}_{fref} + \frac{V_s \cdot R_f}{L_f} \cdot I_{df} + \frac{V_s (V_{dfeq} + V_{dfn})}{L_f} + w_s \cdot V_s \cdot I_{df} - \frac{V_s^2}{L_f} \\ \dot{\mathcal{S}}_2 = \dot{Q}_{fref} - \frac{V_s \cdot R_f}{L_f} \cdot I_{df} - \frac{V_s (V_{dfeq} + V_{dfn})}{L_f} + w_s \cdot V_s \cdot I_{df} \end{cases} \quad (37)$$

In the sliding and permanent mode, we obtain the expression of the equivalent command supplied as follows:

$$\begin{cases} V_{dfeq} = \frac{L_f}{V_s} \cdot \dot{\mathcal{S}}_1 - R_f \cdot I_{df} + L_f \cdot w_s \cdot I_{df} \\ V_{dfeq} = -\frac{L_f}{V_s} \cdot \dot{\mathcal{S}}_2 - R_f \cdot I_{df} - L_f \cdot w_s \cdot I_{df} + V_s \end{cases} \quad (38)$$

The stabilizing control is expressed by :

$$\begin{cases} V_{dfn} = K_{dfn} \text{sat}(e_4) \\ V_{qfn} = K_{qfn} \text{sat}(e_3) \end{cases} \quad (39)$$

6. DC-BUS VOLTAGE CONTROL LOOP

The DC-BUS equation can be expressed as indicated [21]:

$$\begin{cases} W_c = \int P_c \cdot dt = \frac{1}{2} \cdot C \cdot U_{dc}^2 \\ \frac{dU_{dc}^2}{dt} = \frac{2}{C} \cdot (P_f - P_r) \end{cases} \quad (40)$$

With: W_{dc} : the DC- Bus energy, and P_c DC- Bus power. By neglecting all the losses exchanged between the electrical grid and the rotor of the DFIG, the powers involved on the DC-BUS can be written as follows:

$$\begin{cases} P_f = P_c + P_r \\ P_c = U_{dc} I_c \end{cases} \quad (41)$$

We can control the power P_c in the capacitor by regulating the power P_f , and thus to adjust the DC-BUS voltage.

7. RESULTS AND DISCUSSION

In order to gauge the performance of the wind system based on the Doubly Fed Induction Generator, we tested the operation of the wind system by two types of control: the linear control by Flux Oriented Control Technique and the nonlinear control by sliding mode technique to control the powers generated by the DFIG and also to control the DC-BUS Voltage. To carry out this work, two tests were performed:

- Test 1: Tracking and regulation tests for SMC and PI.
 - Test 2: The tests of robustness regarding the variation parameters.
- The results obtained for the various simulation tests, are respectively exposed on the Figure:

- Figure 11 presents the tracking and regulation tests for SMC and PI.
 - Figure 12 presents the tests of robustness regarding the variations parameters.
- By applying a wind profile given by (Figure.a), we can observed the performance of the turbine shown in Figure 10.

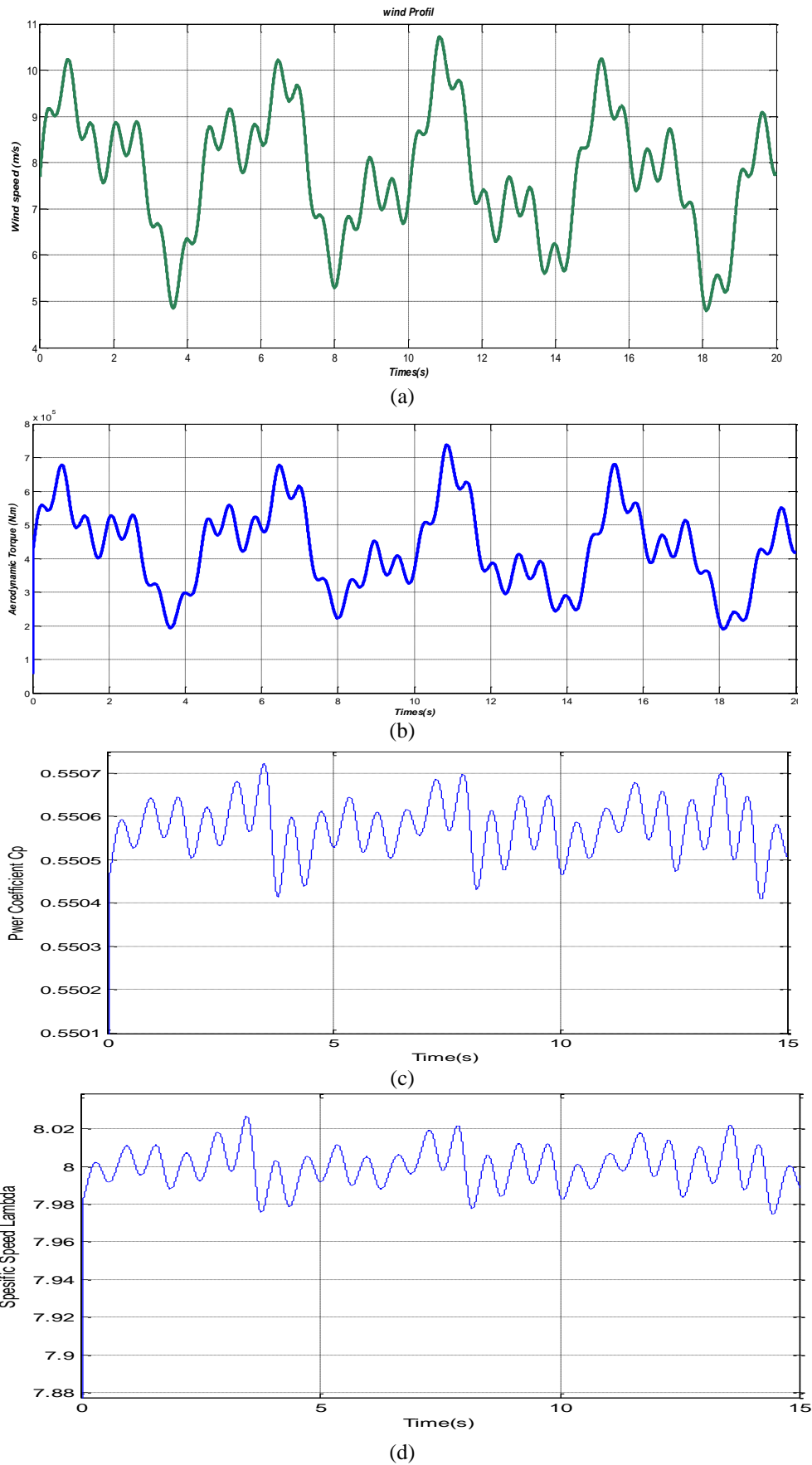


Figure 10. the performances of wind turbine according to the MPPT strategy

From these Figures, we can see that the aerodynamic power according to the MPPT (b) has the same form as that of the wind profile. We can also see that the power coefficient (c) and the specific speed lambda (d) reach their maximum values which ensure that the wind system can provide an optimal power from the wind.

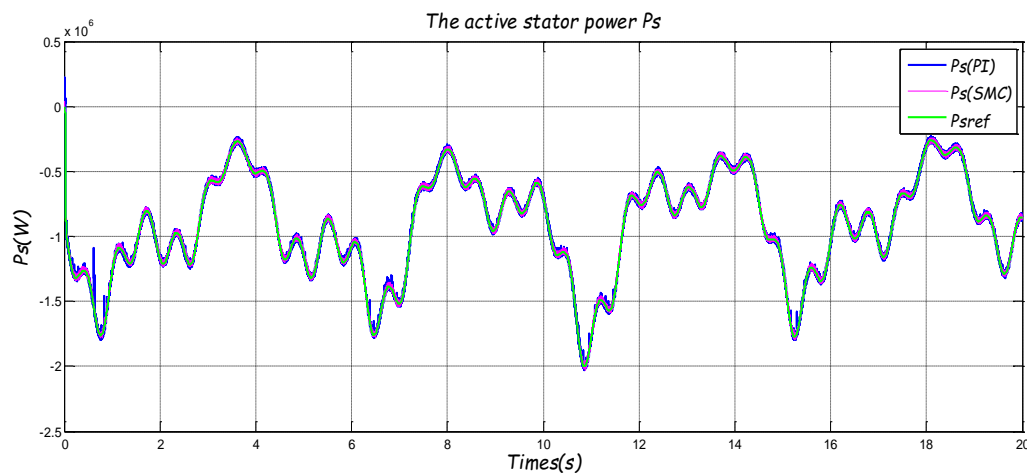
7.1. Tracking Tests

In this case, we consider the aerodynamic power according to the MPPT strategy as a reference for the stator active power of the DFIG and zero as a reference of the stator reactive power to guarantee a unit power factor on the stator side to optimize the quality of the energy reverted on the grid. By these Figures, we demonstrate the active and reactive power variations.

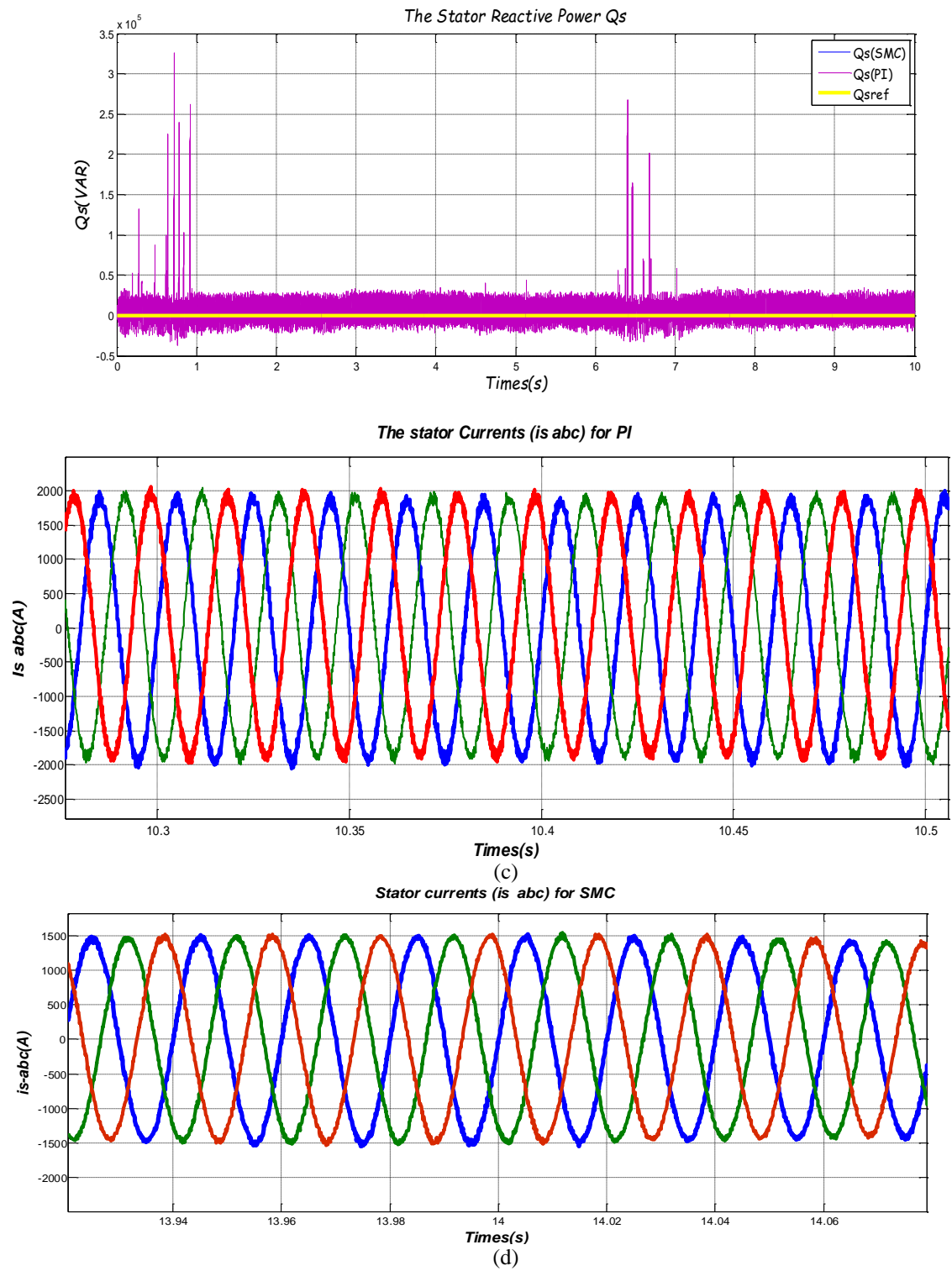
From Figure 11 (Figure(a) and Figure (b)), the references power are well followed for both controllers (PI and SMC). However, we can see the advantage of the Sliding Mode SM approach presented in the short response time ($T_{PI}=0.045s$, $T_{SMC}=0.003s$) and the perfect tracking of their reference compared to the PI controller. The stator active power (Figure .a) is negative $P_s < 0$ means that the grid receives the energy produced from the DFIG. The stator currents I_{s-abc} (Figure.c and Figure.d), have a sinusoidal shape for both controllers but with an improvement in quality for the SMC. From Figure.e, we can observe that the DC-BUS voltage follows perfectly its reference for both controllers but with a very short response time, and less undulations for the Sliding Mode Controller.

7.2. Robustness Tests

The previous test was done considering the fixed machine parameters, but these parameters can be influenced by several physical phenomena such as the temperature which allows the increase of the resistance values, the saturation of the inductors ... etc. In addition, the identification of these parameters is expressed in infidelities owing to measuring devices and the adopted methodology. Therefore, it is interesting to compare the performance of both systems regarding this phenomenon. To test the performance of each controller against model uncertainties affecting system stability, we replace the parameters of the system used in the DFIG by the following equation: $\begin{cases} R_{(s,r)} = 2R_n, L_{(s,r)} = 1.2L_n \end{cases}$



(a)



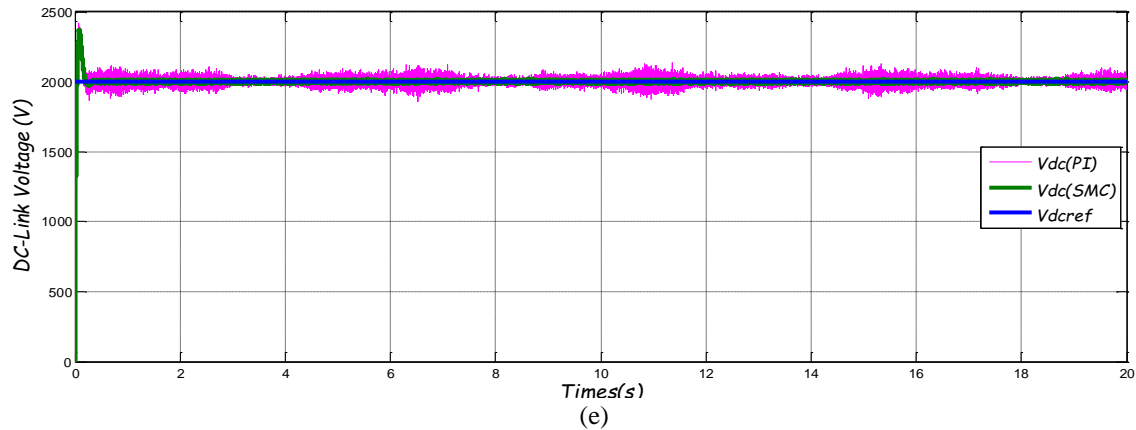
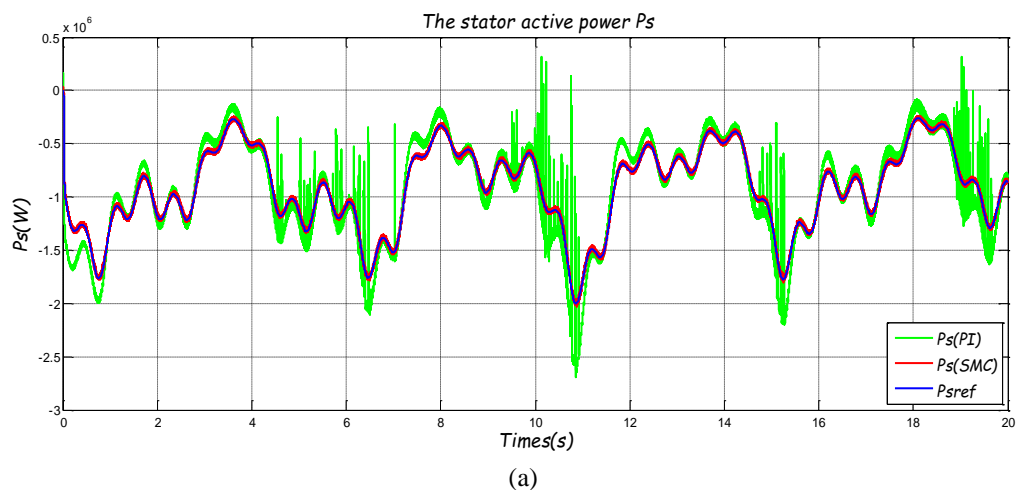
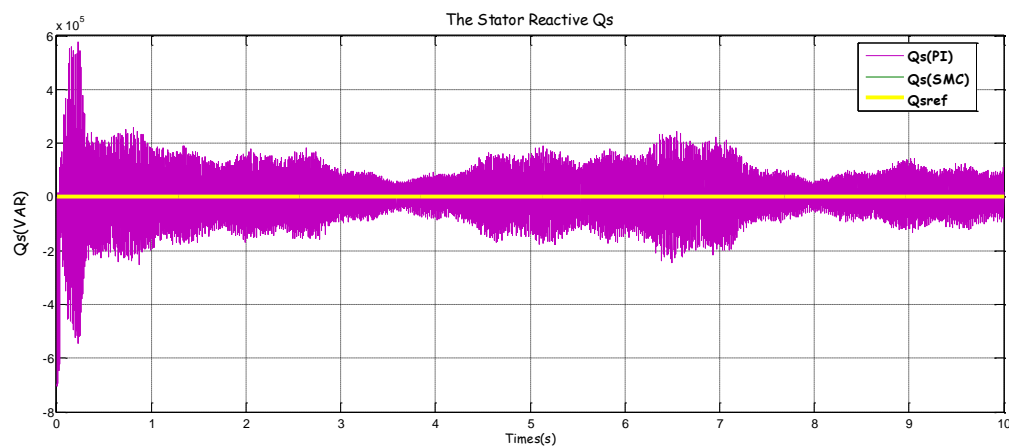


Figure 11. Tracking test for SMC and PI

From Figure.12, we can note that the SMC controller is not influenced by the parametric variations compared to the PI controller where we see a slight variation and a response time that has been significantly increased. Which confirms the Sliding Mode Technique robustness regarding the system parametric variation.



(a)



(b)

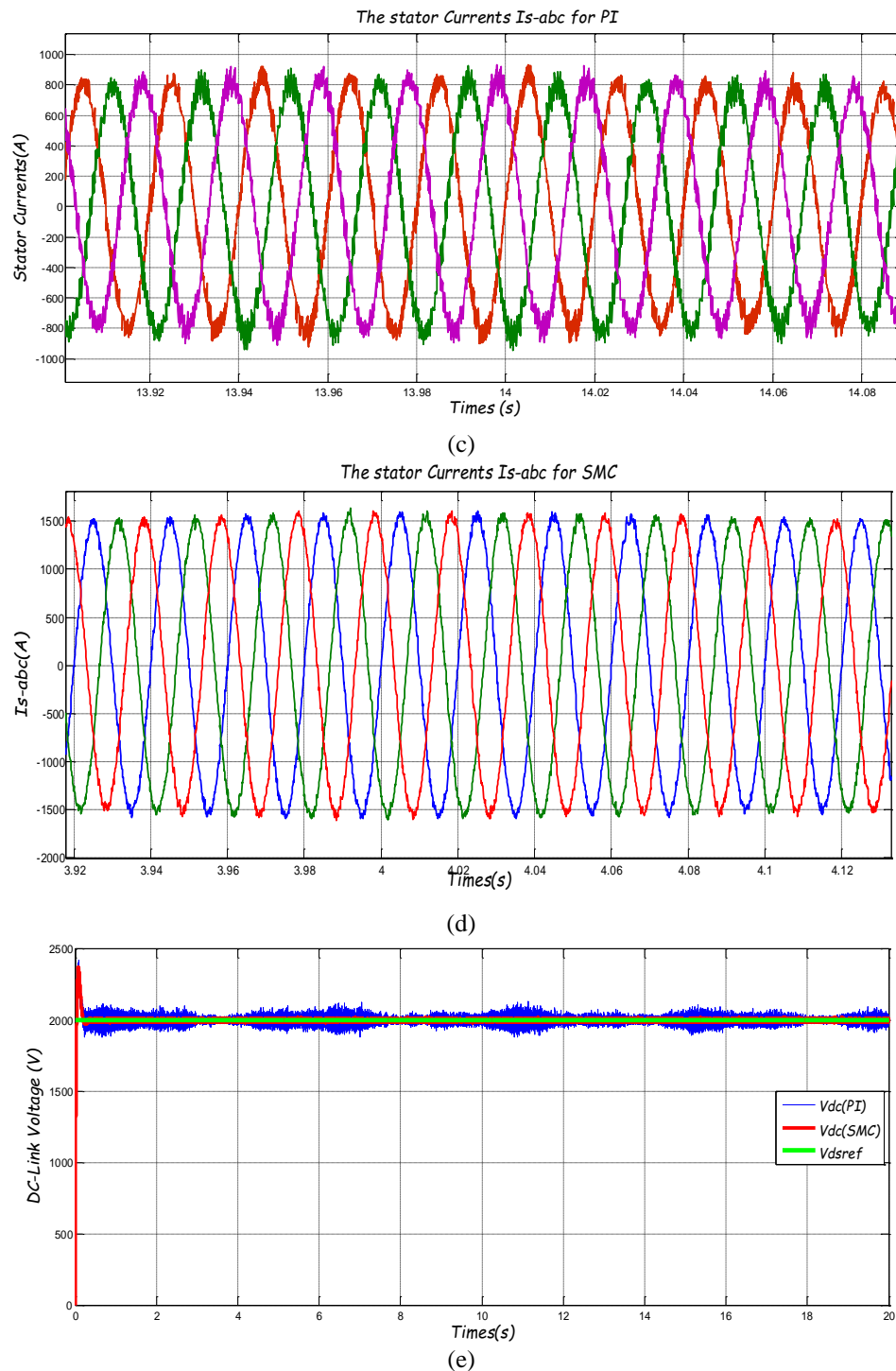


Figure 12. Robustness tests for SMC and PI

8. CONCLUSION

This work presents a comparative analysis between the Sliding Mode Technique and the Flux Oriented Control using PI controller applied to the wind system based on a Doubly Fed Induction Generator DFIG in order to control the Back-to-back converters (RSC and GSC) with the objective of extracting and transmitting the maximum of the wind power to the grid. Firstly, we modeled the turbine, the DFIG as well as the RSC and GSC converters in order to apply both control techniques PI and SMC. Secondly, we used the MPPT control technique to provide the active power reference for our control block, while the reactive power is kept equal to zero, in order to keep a unit power factor. The simulation results present the performances of

the proposed algorithms, in terms of monitoring and robustness against parametric variations. We performed two simulations for each controller. We found that the SMC provides more powerful and meaningful results compared to the PI controller and therefore this technique presents a good contender for controlling DFIG integrated in a wind power system.

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